

65 YEARS EXPERIENCE
WITH CASE HARDENED
& GROUND HIGH SPEED
GEARS

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1 Introduction

This paper discusses the development, and particularly the reliability, of MAAG speed increasing gears which drive turbo compressors of high speeds and powers. An important characteristic of these gears is the case-carburized, hardened and ground gear teeth. The teeth are geometrically corrected to compensate for deflections due to load and thermal effects.

Highest priority has to be given to the assurance of the reliability of such gears. Any breakdown of gears not only may result in large production losses, but also entails the danger of consequential damage to other components of the gear or even to other machinery. Furthermore, many of these gears are installed in locations of difficult access, such as: deserts; arctic regions; or platforms, which may turn a minor problem into a major logistic operation. Based on decades of experience with these gears, a reliability analysis is performed on them and presented in this paper.

2 The Development of the Case-carburized, Hardened and Ground Turbo Gear

The first gear of this type was designed and built by MAAG in 1919 and was installed between a steam turbine and a compressor. At this time surface hardening of steel by carburization had generally been practiced for quite some time. This technique, however, could only be applied to gears that had to operate at low speeds, because of the relatively large distortions that this kind of heat treatment caused.

The engineer Max Maag designed and built the first gear tooth grinding machine in 1916. It enabled the production of accurate hardened gears for every required speed. Max Maag could not know at that time, that his pioneering work would eventually pave the way for the carburized case hardening technique to become superior to all other heat treatment techniques of gears as far as load capacity and wear resistance are concerned.

MAAG, the company Max Maag founded, has improved this technique ever since and developed carburization into an automatic and highly reliable process. More than 65 years of experience with these kinds of gears has

been accumulated in the meantime. Today this process is used throughout the machine industry, in aircrafts, for ship propulsion gears, traction, steel and cement mills, etc. Design engineers were often hard to be convinced about the superiority of the hardened tothing. Decades of patient work were necessary to achieve the breakthrough, as reflected in Table 1.

MAAG's development of the turbo gear, of course, profited from the experience gained in all these other fields. It was a process of small steps towards higher speeds and transmitted power. Operating experience, and knowledge gained from the careful investigation of equipment that had failed, was fed back to the design office. For quite some time, the process of "learning by mistakes" was adequate to maintain the leading position in this field.

With increasing pitch line velocities, however, the tooth contact deteriorated due to thermal distortions of the rotors. Temperature measurements, performed experimentally under operating conditions, helped us to evaluate the optimum thermal tooth correction.

New test gears are being built and will be tested at loads and speeds anticipated for the immediate and distant future. The aim of these tests is to be able to calculate compensating tooth corrections accurately for any new application.

Parallel to this development of the tothing, the bearings needed increased attention as well. The ever increasing demands of bearing loads and speeds and the stringent vibrational requirements were, and are, met by continuous improvements in design, all tested on our own bearing test stand, at loads and speeds well over the intended application.

To this date, MAAG has delivered over 2,000 high speed gears, which transmit powers between 1,500 kW and 65,000 kW and with pitch line velocities close to 200 m/s. These gears have proven to be highly reliable as shall be shown in the following chapter. This high level of reliability has been achieved not only by the great efforts made in research and

development, but also through stringent quality assurance and complete after-sales services, including:

- on site erection
- regular inspection and overhauls
- trouble-shooting analysis, including whole trains, if necessary
- schooling of customer's operating and maintenance staff

Furthermore, MAAG is in the fortunate position to be a worldwide leading manufacturer of machine tools and measuring machines used in the manufacturing and quality assurance of such high speed gears.

3 Reliability

The reliability analysis presented here deals with gears in three large separate groups of turbo compressor installations, each group with its own typical ambient and operational conditions but all with high requirement regarding power, pitch velocity and ratio. Each group consists of 50 to 80 MAAG gear units in operation. This grouping enables an easier and more meaningful judgment than an overall analysis (see fig. 1 and 2).

The purpose of the analysis is to provide data on the reliability of MAAG's gears that is generally and easily understood and, still meaningful.

For more detailed theories on reliability there is a vast amount of literature available. The following terms on reliability shall be defined for the purpose of this presentation:

Failure is assumed to be an event that causes a gear to become inoperable, so that it can only be put back into service after some corrective maintenance has been performed.

Failure rate (λ) represents the number of failures that occur per unit of operating time, at a certain time during the life time of the components.

Mean Time between Failures (MTBF) represents the average number of operating hours, between failures.

Mean Time to Repair (MTTR) represents the average number of hours it took to repair a gear within its group.

Availability is defined as the probability that at any time the gear is in an operating state.

The above terms are mean values, which will be calculated from data over the entire operating time, up to the present, i.e. April, 1986. The cumulated hours of operation of some gears are not known exactly, especially of installations that are not regularly serviced by MAAG. The yearly production for such installations known to be in operation is conservatively estimated at 4,000 hours. Gear failures or major production stops in such installations are reported to us, however, and are therefore considered in this analysis.

MAAG gears in turbo compressor trains for reinjection plants in Algeria

(Group 1)

A total of 57 MAAG speed-increasing gears are installed between gas turbines and compressors. These installations are situated in two areas of the Sahara desert. The total power installed adds up to 1,300,000 kW.

Table 2 gives some performance data of these gears, as well as hours of operation and the cases of failure up to the present time. Details of these failures are provided in Table 3. From this data the following terms are calculated:

$$\text{Failure rate} = \frac{4 \text{ failures}}{1,447,000\text{h}} \times 1,000 = <0.003 \text{ failures per } 1000 \text{ hours}$$

$$\text{MTBF} = \frac{1,447,000}{4} = 360,000 \text{ hours}$$

MTTR = 192 hours per failure

$$\text{Availability} = \frac{\text{MTBF}}{\text{MTBF} + \text{MTTR}} = 0.9994$$

Comments

The MTBF of the gears under consideration is 360,000 hours, i.e. one failure in 360,000 hours of total operation must be expected, or 56 out of 57 gears shall still be in operation after 6,340 hours of operation, one will not.

All four tooth fractures were fatigue failures of the same nature, all on the pinion. In each case, the fatigue crack originated close to the pitch circle on the loaded flank at one or two teeth, slightly to one side from the centre of the toothing. Exact geometrical measurements of the failed pinions and examinations of the steel revealed a metallurgical problem to have been responsible for all four failures. The heat treatment was changed to adapt to the high operating temperatures, which these gears are subjected to. Fortunately, this correction in heat treatment could be introduced in time to cover the bulk of these gears before they were built for these installations. It must be noted, that, when these gears were designed, such high pitch line velocities had never been attempted before for such installations.

No further problems have arisen since the completion of these remedial actions; i.e. since mid 1979 no unscheduled shut-downs were necessary due to the gears. Considering that the four tooth fractures occurred one to five years after start-up, they could be classed as running in failures, i.e. after the elimination of this one problem, reliability and availability of this group of gears has increased sharply.

MAAG gears in turbo compressor trains for off shore oil & gas production platforms

(Group 2)

A large number of MAAG turbo gears are installed offshore, where the demands on reliability are especially exacting. Platforms, with Maag gears installed on them, are scattered all over the world. Although the climate may differ widely among the different sites, other ambient and operating conditions of the gears on platforms are much alike.

Since 1971 MAAG has delivered 123 gears for this type of installation, of which 80 are now in operation, see table 4. Details of the failures that occurred up to the present are listed in table 5.

$$\text{Failure rate} = \frac{11}{2,014,000} \times 1,000 = 0.0055 \text{ failures per 1,000 hours}$$

$$\text{MTBF} = \frac{2,014,000}{11} = 183,090 \text{ hours}$$

$$\text{MTTR} = 91 \text{ hours per failure}$$

$$\text{Availability} = \frac{189,090}{183,090 + 91} = 0.9995$$

Comments

In at least six out of a total of eleven cases of failure, the gear suffered consequential damage, i.e. the defects originated clearly outside of the gear, as for example failures of the lube oil supply. This fact is not reflected in the above figures, but it points towards the importance of a careful and thorough system design, in close collaboration with the main machinery contractor, and the implementation of the resulting maintenance program as recommended by the gear manufacturer.

In three cases, the corrective measures were able to be undertaken during planned overhaul, i.e. they did not cause production losses.

MAAG gears for turbo compressor trains in gas liquification plants in Saudi Arabia

(Group 3)

In 1960 MAAG delivered the first turbo gears for compressors to Saudi Arabia. To date, 82 units with MAAG gears have been installed with a total power of close to one million kilowatts.

The boom time for new installations was 1977/78, when 45 large speed increasers were built with power between 11,765 kW and 23,162 kW, all being driven by synchronous electric motors.

Table 6 gives all pertinent data for these gears. As reliable information, pertaining to operating hours, is only partly available at this time, the year in which the gears were commissioned is given instead.

Table 7 provides a summary of the cases in which failures occurred. This list may be incomplete, because the maintenance work is carried out by the customer, who is well equipped with spare parts. The inspection work is limited to checking the tooth contact annually.

As serious problems would certainly be reported to us, we assume that, to date, no further breakdowns of any kind have occurred. On the other hand, a reliability study, as carried out in the previous chapters is not possible at this time, but it would certainly show similar, or even more favorable figures than in groups 1 and 2.

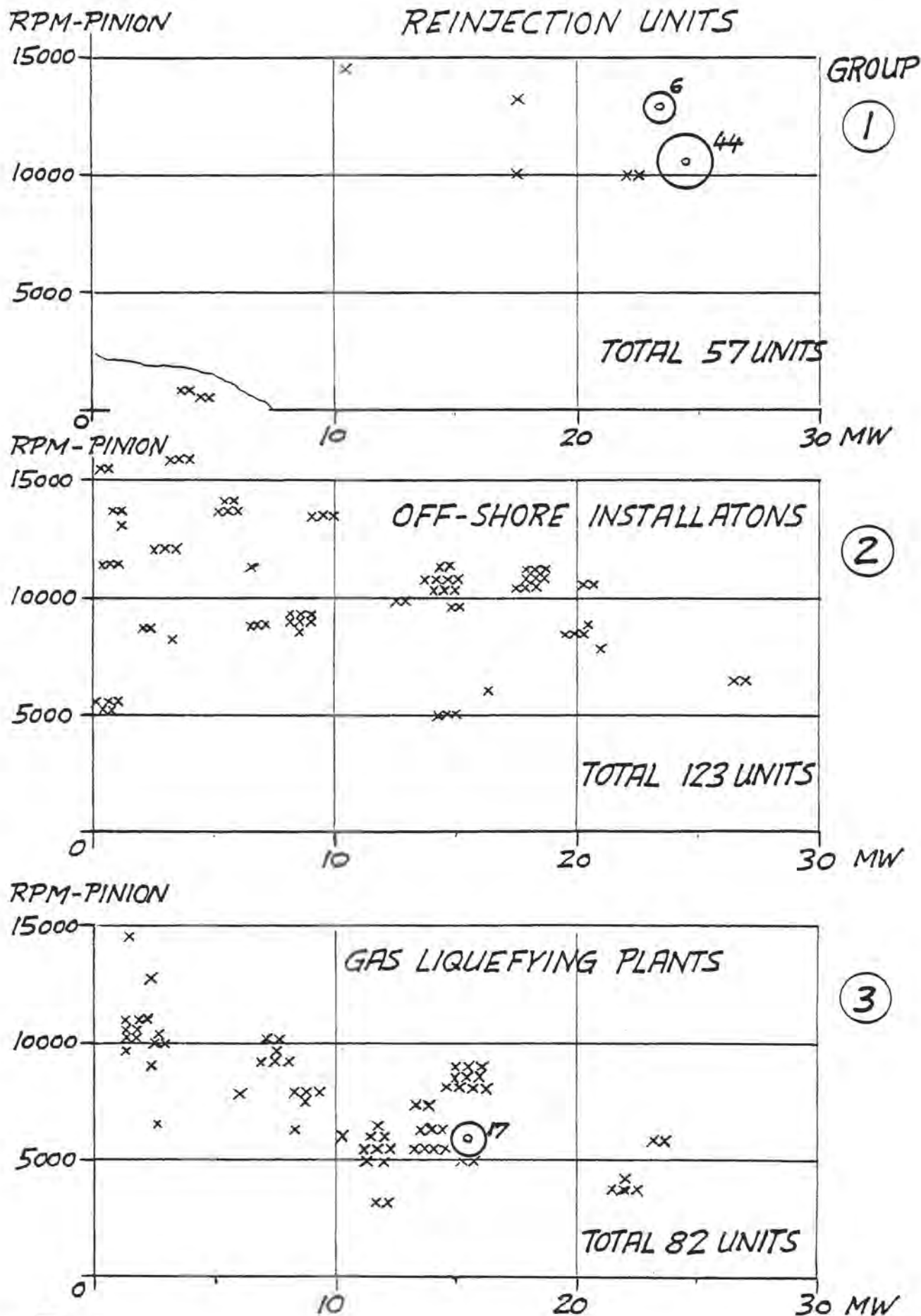


Fig. 1



Fig. 2

Table 1

Milestones in the Development of the Case-Carburized,
Hardened and Ground Gears

1916	First gear tooth grinding machine constructed.
1919	FIRST HARDENED and GROUND TURBO GEAR
1932	First hardened and ground marine gear
1946	TURBO GEAR with PITCH LINE VELOCITY 112 m/s
1949	Construction of gear grinding machine for 3,600 mm diameter
1952	Largest gear measuring machine, 1000 mm diameter
1952	First locked train gears for ship propulsion, 27,500 HP
1956	TURBO GEAR 26,838 kW; 2,988/1,000 rpm; 118 m/s pitch line velocity
1956	First locked train gears for propulsion of tankers
1959	Largest gear grinding machine, 4,600 mm diameter
1960	Development and construction of automatic drip-feed carburization
1964	TURBO GEAR 62,500 kW; 2,988/1,000 rpm; 137 m/s pitch line velocity
1965	First automatic synchronizing clutch, 20 MW/3,000 rpm for power station peak power demand
1966	Epicyclic gear for cement mill, torque 3,000 kNm
1968	First COGOG installation, all gas turbine propulsion for naval crafts
1969	TURBO GEAR 14,560 kW; 5,500/14,837 rpm; 151 m/s pitch line velocity
1969	Hardened gears for largest steel mill in the world
1972	TURBO GEAR 17,500 kW, 5,103/13,132 rpm; 173 m/s pitch line velocity
1974	First CODOG installation, gas turbine/diesel propulsion for naval crafts
1975	Hardened high precision gear drives for the largest telescope in the southern hemisphere (ESO)
1980	Largest gear measuring machine; 2,000 mm diameter
1981	Largest epicyclic gear for cement mill, torque 5,683 kNm
1983	TURBO GEAR 55,000 kW; 6,340/3,000 rpm; 153.5 m/s pitch line velocity
1984	TURBO GEAR (ALASKA) 34,579 kW; 4,670/10,720 rpm, 173 m/s
1986	TURBO GEAR 60,000 kW; 6,340/3,600 rpm; 173 m/s pitch line velocity

Table 2. Reinjection units Algeria

Power kW	Revs. RPM	PLV m/s	Number of Units	Hours of operation by April 1986		Casenum- ber of Failure
				per Unit	Total	
23 934	4670/10 335	158	6	50 000	300 000	3
24 670	4670/10 335	158	5	30 000	150 000	
24 670	4670/10 335	158	20	25 000	500 000	
24 670	4670/10 335	158	11	20 000	220 000	
24 670	4670/10 335	158	2	1 000	2 000	
23 300	4670/12 924	169	6	5 000	30 000	
17 500	5103/13 132	173	1	65 000	65 000	1
17 500	5103/10 070	160	1	65 000	65 000	2
22 365	3600/10 067	153	1	50 000	50 000	
22 365	3600/10 067	153	1	50 000	50 000	4
10 570	6500/14 589	151	3	5 000	15 000	
1 332 300			57		1 447 000	Total

Table 4. Off-shore installations

Power kW	Rated Speed RPM	PLV m/s	Number of Units	Hours of operation by April 1986		Casenum- ber of Failure
				per Unit	Total	
16 300	6057/3 /Gener.	105	1	100 000	100 000	1
14 500	5000/10 377	141	3	46 000	138 000	
8 456	1792/ 8 819 E-Mot.	94	1	50 000	50 000	
6 985	1792/ 8 819	94	6	50 000	300 000	2, 3
19 853	3429/ 8 467	128	3	26 000	78 000	4
9 450	6540/13 578	143	3	40 000	120 000	
2 050	1787/ 8 744	56	2	30 000	60 000	
850	8744/13 711	101	2	30 000	60 000	
5 472	9536/13 650	147	5	40 000	200 000	5
3 603	7500/15 849 /21 538	131	3	36 000	108 000	6, 7
6 830	6190/ 8 863	136	3	35 000	105 000	
18 250	3600/10 895	142	3	20 000	60 000	
641	1788/11 372	40	3	20 000	60 000	
17 892	5250/10 619	147	3	20 000	60 000	8, 9
14 200	5000/10 300	127	2	7 000	14 000	
14 816	5200/ 9 609	127	2	28 000	56 000	
3 890	7950/18 972	147	2	10 000	20 000	
3 000	2975/12 092	80	3	10 000	20 000	
14 200	5000/10 521	128	2	15 000	30 000	
4 400	7950/18 754	146	2	14 000	28 000	
220	3580/15 410	49	2	15 000	30 000	
14 630	5004/ 1 800	97	3	14 000	42 000	
26 500	6490/ 3 000	122	2	13 000	26 000	
20 915	3600/ 7 870	122	1	20 000	20 000	
3 200	2970/ 8 188	82	1	12 000	12 000	
20 450	6190/ 8 826	152	1	10 000	10 000	
6 580	1485/11 296	96	1	10 000	10 000	
373	5200/ 1 803	27	5	15 000	75 000	
12 530	5081/ 9 781	126	2	7 000	14 000	
18 250	3600/10 895	142	3	6 000	18 000	
14 775	6178/11 350	151	2	18 000	36 000	
20 220	5200/ 9 672	153	2	18 000	36 000	
1 100	2966/15 708	63	1	18 000	18 000	

786 135			80		2 014 000	Total in Service
575 779			43		0	Total delivered but not yet in service
1 361 914			123	Total delivered		

Table 6. Liquification plant Saudi Arabia

Power kw	Rated		Number of Units delive- red	Year of Commissio- ning (approx.)	Total hours of operation	Casenum- ber of Failure
	Speed RPM	PLV m/s				
1 306	1730/10 107	56	4	1961		
2 550	4405/ 9 881	90	1	1961		
22 000	2990/ 3 600	107	3	1965	700 000	1, 2, 3
1 500	6440/14 446	103	1	1965		
2 600	6500/ 1 782	53	1	1965/67		
2 575	4440/ 9 952	80	2	1967		
8 750	5100/ 7 465	127	1	1969		
8 460	1800/ 7 686	92	3	1977		
2 310	1788/ 9 055	56	1	1976		
1 200	1785/ 9 789	51	1	1976		
1 200	1785/10 294	51	1	1976		
13 970	1800/ 6 146	102	3	1977		
13 970	1800/ 5 523	100	4	1976/77		
1 850	1792/10 999	58	1	1977		
8 240	1800/ 6 214	88	1	1977		
15 441	1800/ 8 652	116	6	1980/81	150 000	
15 441	1800/ 8 691	116	4			
15 441	1800/ 4 995	100	2			
15 441	1800/ 5 811	104	9	1980	225 000	5
15 441	1800/ 5 886	104	8	1981	152 000	
11 765	1800/ 5 312	99	6	1980		
11 765	1787/ 5 409	99	2	1980		
13 235	1786/ 7 348	105	2	1980		
5 966	1782/ 7 900	76	1	1982		
23 162	1200/ 5 833	102	2			
10 294	1787/ 6 026	104	1	1982		
11 765	1785/ 3 106	86	2	1982		
7 083	1785/10 133	95	2	1982		
1 850	1792/10 999	58	1	1982		
7 463	1800/ 9 218	95	4	1980		
2 238	3575/12 751	82	1	1982		
22 000	2990/ 3 600	107	1	1982		
954 343			82			Total

